

Soln:

Suppose, the infinite line through A is a source which emits fluid at the rate of $2\pi m \rho$ units of mass per unit length per unit time.

This is the mass flux per unit time per unit length of the infinite cylinder whose trace in the plane of flow is the closed curve γ and m is the strength of the line source.

Now, let us take γ_0 to be the circle centred at A with radius a .

Since the speed of blow \vec{q} is same everywhere on γ_0 . The mass flux is $2\pi a \rho q$ per unit length.

$$2\pi m \rho = 2\pi a \rho q$$

$$\Rightarrow q = m/a$$

The velocity potential on such a

circle is given by,

$$\vec{q} = -\nabla \phi$$

where, $\phi = \phi(r)$

$$m/r = -\frac{\partial \phi}{\partial r}$$

$$m \frac{\partial r}{\partial r} = -\partial \phi$$

$$\partial \phi = -m \frac{\partial r}{\partial r}$$

$$\phi = -m \log r.$$

If we consider the point P on the circle

having polar co-ordinates $P(r, \theta)$ w.r.t.

'A' as origin and if $\overline{OP} = z = re^{i\theta}$ then,

$$\log r = \log |z|.$$

$$\phi = -m \log r$$

$$\therefore \phi = -m \log |z|.$$

Hence we see that the complex velocity

potential of the line source through A of

uniform strength m is given by,

$$w = -m \log z \Rightarrow z = re^{i\theta} \Rightarrow \log z = \log(r e^{i\theta}) \\ = \log(r) + i\theta$$

$$\omega = -m \{ \log(r) + i\theta \}$$

$$\psi(r, \theta) = -m\theta$$

$$\phi(r, \theta) = -m \log r$$

The stream lines $\psi = \text{constant}$ shows that in any plane of flow all the straight lines

$$\theta = \text{constant}$$

The equi-potential $\psi = \text{constant} \Rightarrow r = \text{constant}$

which are concentric circles centered on origin.

NOTE :

If the line source of strength m .

where, situated at $z=z_1$, instead of at $z=0$

then $r = |z-z_1|$ and the complex velocity

potential becomes

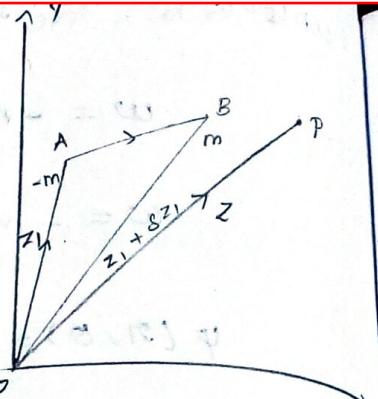
$$w = -m \log |z-z_1|$$

$$w = -m \log(z-z_1)$$

LINE DOUBLETS :-

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Figure shows a uniform line source of strength m at A and $-m$ at B and P be the fixed point all in the same plane of flow.



$$\text{radius } \overline{OA} = z_1, \overline{OB} = z_1 + \delta z_1, \overline{OP} = z$$

$$\overline{AP} = \overline{OP} - \overline{OA}$$

$$\therefore \overline{AP} = z - z_1$$

$$\overline{BP} = \overline{OP} - \overline{OB}$$

$$\therefore \overline{BP} = z - z_1 - \delta z_1$$

The complex potential at P due to the two line sources at A, B is,

$$w = -m \log(z - z_1)$$

$$w = -[-m \log(z - z_1) + m \log(z - z_1 - \delta z_1)]$$

$$w = m \log(z - z_1) - m \log(z - z_1 - \delta z_1)$$

①

using Taylor's expansion,

$$\log(z - z_1 - \delta z_1) = \log(z - z_1) - \delta z_1$$

$$\frac{\partial}{\partial z} [\log(z - z_1)] + O(\delta z_1^2)$$

$$\log(z - z_1 - \delta z_1) = \log(z - z_1) - \delta z_1 \frac{1}{z - z_1} + O(\delta z)^2 \rightarrow ②$$

Here we assume that $|\delta z_1|$ is very small

that higher powers $|\delta z_1|^2, |\delta z_1|^3, \dots$ all negligible.

\therefore Approximating to the 1st order we have,

using ② in ①,

$$w = m \log(z - z_1) - m \left\{ \log(z - z_1) - \delta z_1 \frac{1}{z - z_1} \right\}$$

$$= m \log(z - z_1) - m \log(z - z_1) + m \delta z_1 (z - z_1)^{-1}$$

$$w = m \delta z_1 (z - z_1)^{-1} \rightarrow ③$$

If AB makes an angle α with Ox then,

$$w = m |\delta z_1| e^{i\alpha} (z - z_1)^{-1}$$

suppose that m becomes very large and

$|\delta z_1|$ is very small in such a way that,

$$\mu = m |\delta z_1|$$

remains finite and constant.

Then the equal and opposite line sources at A are said to form a line doublet of strength μ per unit length \overline{AB} giving the direction of the axis of the doublet.

Hence the complex potential at P is given by,

$$w = \mu e^{i\alpha} (z - z_1)^{-1}$$

PROBLEM: 3:

Discuss the flow due to the uniform line

doublet at 'o' of strength μ per unit length.

It's axis being along ox .

Solution:

The complex velocity potential at $P(x, y)$ is, doublet $\Rightarrow w = \mu e^{i\alpha} (z - z_1)^{-1}$

$$w = \frac{\mu}{z} = \frac{\mu e^{i\alpha}}{(z - z_1)} \Rightarrow \alpha = 0 \quad z_1 = 0$$

$$\text{axis lies along } ox \Rightarrow \frac{\mu \cdot e^0}{z} = \frac{\mu}{z}$$

$$w = \frac{\mu}{x+iy}$$

$$= \frac{\mu}{x+iy} x \frac{x-iy}{x-iy}$$

$$= \mu \left[\frac{x-iy}{x^2+y^2} \right] \Rightarrow (x+iy)(x-iy) = (x^2+y^2)$$

$$\operatorname{Re} w = \phi(x, y) = \frac{\mu x}{x^2+y^2}$$

$$\operatorname{Im} w = \psi(x, y) = -\frac{\mu y}{x^2+y^2}$$

The equipotentials are given by,

$$\phi = \text{constant} \Rightarrow C = \frac{\mu x}{x^2+y^2} \Rightarrow x^2+y^2 = \frac{\mu x}{C}$$

$$x^2+y^2 = 2K_1 x$$

$$(ie) x^2+y^2 = 2K_1 x.$$

which are co-axial circles having centre along the x -axis.

Similarly,

The stream lines $\psi = \text{constant}$ are given by,

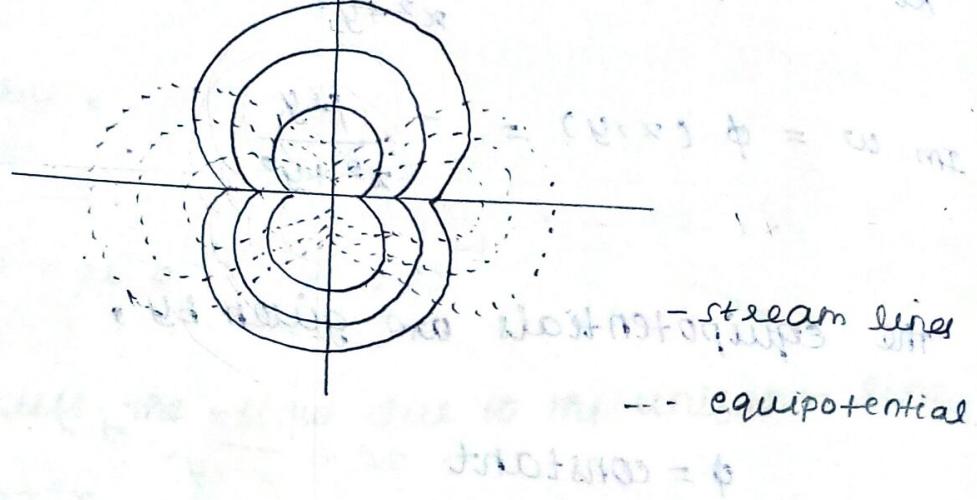
$$x^2+y^2 = 2K_2 y \Rightarrow C = \frac{-\mu y}{x^2+y^2} \Rightarrow x^2+y^2 = \frac{-\mu}{C} y$$

$$x^2+y^2 = 2K_2 y.$$

which are again co-axial circles.

The first family has centre (x_1, k_1)
 radii k_1 . The
 and second family has centre $(0, k_2)$ and
 radius k_2 .

The two families are mutually orthogonal.



LINE VERTICES :-

FOR A TWO DIMENSIONAL FLOW,

$$\vec{q} = u\vec{i} + v\vec{j}$$

where $u = u(x, y)$ and $v = v(x, y)$

then the velocity vector is given by $\vec{q} = \nabla \phi$

$$\Rightarrow \vec{q} = \left(\frac{\partial \phi}{\partial x} - \frac{\partial u}{\partial y} \right) \vec{k} \Rightarrow \vec{k} = \begin{vmatrix} i & j & k \\ \frac{\partial \phi}{\partial x} & \frac{\partial \phi}{\partial y} & \frac{\partial \phi}{\partial z} \\ u & v & w \end{vmatrix}$$

this shows that in two dimensional flow
the vorticity vector is \perp to the plane of flow.

PROBLEM 4:

discuss the 2-dimensional flow for which
the complex potential is

$$\omega = \left(\frac{i\kappa}{2\pi} \right) \log z,$$

where κ is a real constant.

solution:-

put $z = re^{i\theta}$ in the given equation of ω

$$\omega = \frac{i\kappa}{2\pi} \log(re^{i\theta})$$

$$= \frac{i\kappa}{2\pi} [\log r + i\theta].$$

$$\omega = \frac{i\kappa}{2\pi} \log r - \frac{\kappa\theta}{2\pi}$$

equating the real and imaginary part.

we have,

$$\phi = \left(-\frac{\kappa}{2\pi} \right) \theta.$$

$$\Psi = \left(\frac{k}{2\pi} \right) \log r$$

$\phi = \text{constant}$

$\Rightarrow \theta = \text{constant}$ (con) the equi-potential are the radial vector to the origin.

thus,

$\psi = \text{constant}$

$\Rightarrow r = \text{constant}$

(i.e) the stream lines in the plane of flow

of concentric circles $r = \text{constant}$ and the two families are mutually orthogonal

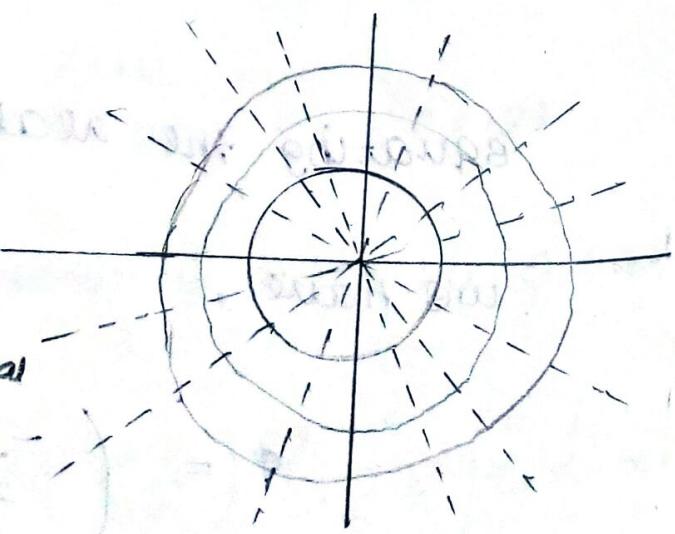
and both ϕ and ψ are harmonic functions.

$r = \text{constant}$

- stream lines

$\theta = \text{constant}$

--- equi potential



The radial and transverse components of the velocity are given by,

$$q_r = - \frac{\partial \phi}{\partial r}, \quad q_\theta = - \frac{1}{r} \frac{\partial \phi}{\partial \theta}$$

$$\Rightarrow q_r = 0 \quad ; \quad q_\theta = \left(\frac{k}{2\pi r} \right)$$

The circulation Γ around any closed curve C surrounding the origin and in the plane

of flow is, given by,

$$\Gamma = \oint \vec{q} \cdot d\vec{s}$$

$$\oint \vec{q} \cdot d\vec{s} \left(\frac{2\pi}{r} \right) = \infty$$

$$\text{Taking } \vec{q} \text{ to the } \vec{q} = q_r \hat{r} + q_\theta \hat{\theta}$$

$$= 0 + \left(\frac{k}{2\pi r} \right) \hat{\theta}$$

$$\vec{q} = \left(\frac{k}{2\pi r} \right) \hat{\theta}$$

$$d\vec{s} = dr \hat{r} + r d\theta \hat{\theta}$$

$$\vec{q} \cdot d\vec{s} = \frac{k r}{2\pi r} dr = \frac{k}{2\pi} dr$$

$$\Gamma = \oint \frac{K}{2\pi} d\theta$$

in plane of flow.

$$K = \frac{k}{2\pi} = \frac{1}{2\pi} \cdot \frac{1}{r} = r\phi$$

$$\Gamma = K$$

If Γ does not surround 'o' (origin), then Γ is zero.

we have shown that a 2-dimensional distribution having a complex velocity potential,

$$\omega = \left(\frac{ik}{2\pi} \right) \log r, \text{ where } k \text{ is a real constant}$$

gives a circulation Γ around any closed curve Γ in the plane of flow and enclosed in the origin 'o' and amount of K .

Also, any other curve in the plane of flow which does not enclose 'o', the circulation is zero.

we now defined a uniform line vortex
as follows,

"It is a uniform distribution along an
infinite line such that the circulation around
any curve γ .

In any plane \perp to that line is constant
to K .
when γ encloses the multiplication of the line
and plane and its zero. when γ does not
contain the intersection."

The strength of uniform line vortex
is defined to be K and its complex velocity
potential is,

$$\left(\frac{ik}{2\pi}\right) \log z, \text{ where the origin is}$$

taken as multiplication of the plane with the
line.

NOTE:

when the line intersects at $z = z_0$,

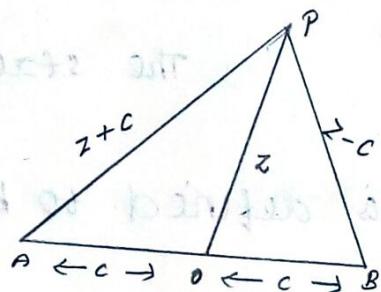
the complex potential is given by,

$$\omega = \left(\frac{iK}{2\pi} \right) \log(z - z_0).$$

PROBLEM : 5:

Find the equations of the stream lines due to uniform line sources of strength m through the points, $A(-c, 0)$, $B(c, 0)$ and a uniform line sink of strength $-m$ through the origin.

Solution:



Let P be the point such that

$$\overline{OP} = z = x + iy.$$

$$\therefore \overline{AP} = z + c$$

$$\overline{BP} = z - c$$

so the complex potential ω due to the

Sources A, B and the sink through the origin in the region is given by,

$$\omega = -m \log(z - z_1)$$

$$\omega = -m \log(z + c) - m \log(z - c) + 2m \log z$$

$$(a+b)(a-b) = a^2 - b^2$$

$$= -m \log(z^2 - c^2) + m \log z^2$$

$$= m [\log z^2 - \log(z^2 - c^2)]$$

$$= m \log \frac{z^2}{z^2 - c^2} \quad (\because \log m - \log n = \log m/n)$$

$$= m \log \frac{(x+iy)^2}{(x+iy)^2 - c^2}$$

$$\omega = m \log \left(\frac{x^2 - y^2 + i2xy}{x^2 - y^2 - c^2 + i2xy} \right)$$

$$= m \log \left(\frac{x^2 - y^2 + i2xy}{x^2 - y^2 - c^2 + i2xy} \right) x$$

$$\left(\frac{x^2y^2 - c^2 - i2xy}{x^2 - y^2 - c^2 - i2xy} \right)$$

$$= m \log \left(\frac{(x^2 - y^2)(x^2 - y^2 - c^2) - i(x^2 - y^2) 2xy + i2xy(x^2 - y^2 - c^2) + 4x^2y^2}{(x^2 - y^2 - c^2)^2 + 4x^2y^2} \right)$$